Technical Report ARAEW-TR-06019

PRESTRESSED CARBON FIBER COMPOSITE OVERWRAPPED GUN TUBE

Andrew Littlefield Edward Hyland

October 2006



ARMAMENT RESEARCH, DEVELOPMENT AND ENGINEERING CENTER
Armaments Engineering & Technology
Weapon Systems & Technology

The views, opinions, and/or findings contained in this report are those of the author(s) and should not be construed as an official Department of the Army position, policy, or decision, unless so designated by other documentation.

The citation in this report of the names of commercial firms or commercially available products or services does not constitute official endorsement by or approval of the U.S. Government.

Destroy this report when no longer needed by any method that will prevent disclosure of its contents or reconstruction of the document. Do not return to the originator.

Form Approved REPORT DOCUMENTATION PAGE OMB No. 0704-0188 Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden to Washington Headquarters Service, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Project (0704-0188) Washington, DC 20503. PLEASE DO NOT RETURN YOUR FORM TO THE ABOVE ADDRESS. 1. REPORT DATE (DD-MM-YYYY) 2. REPORT TYPE 3. DATES COVERED (From - To) 26-09-2006 **TECHNICAL** 4. TITLE AND SUBTITLE 5a. CONTRACT NUMBER PRESTRESSED CARBON FIBER COMPOSITE OVERWRAPPED **5b. GRANT NUMBER** 5c. PROGRAM ELEMENT NUMBER 6. AUTHOR(S) **5d. PROJECT NUMBER** Andrew Littlefield and Edward Hyland 5e. TASK NUMBER 5f. WORK UNIT NUMBER 7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) PERFORMING ORGANIZATION REPORT NUMBER U.S. Army ARDEC ARAEW-TR-06019 Benet Laboratories, RDAR-WSB Watervliet, NY 12189-4000 9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) 10. SPONSOR/MONITOR'S ACRONYM(S) U.S. Army ARDEC Benet Laboratories, RDAR-WSB Watervliet, NY 12189-4000 11. SPONSORING/MONITORING **AGENCY REPORT NUMBER** 12. DISTRIBUTION AVAILABILITY STATEMENT Approved for public release; distribution is unlimited. 13. SUPPLEMENTARY NOTES 14. ABSTRACT The emphasis on lightweight large caliber weapons systems has placed the focus on the use of advanced composite materials. Using composite materials not only directly removes weight from the gun tube but, by better balancing the tube, allows the use of smaller drive systems, thus enhancing the weight loss. Additionally the use of high stiffness composites helps with pointing accuracy and alleviates the dynamic strain phenomenon encountered with high velocity projectiles. Traditionally there were two issues with composite jackets: the coefficient of thermal expansion mismatch between the steel substrate and the composite jacket causing a gap, and the lack of favorable prestress in the jacket. Dealing with these issues greatly complicated the manufacturing process to the point where massproducing the barrels would have been problematic at best. By using a thermoplastic resin & winding under tension the manufacturability of the barrels has been greatly improved, the gap has been eliminated, and a favorable prestress has been achieved. 15. SUBJECT TERMS Carbon fiber composite; composite materials; polyetheretherketone (PEEK); lighweight; large caliber gun tube.

17. LIMITATION OF

ABSTRACT

c. THIS PAGE

18. NUMBER

OF PAGES

16. SECURITY CLASSIFICATION OF:

b. ABSTRACT

a. REPORT

U/U

19a. NAME OF RESPONSIBLE PERSON

19b. TELEPONE NUMBER (Include area code)

Andrew Littlefield

518-266-3972

INSTRUCTIONS FOR COMPLETING SF 298

- **1. REPORT DATE.** Full publication date, including day, month, if available. Must cite at lest the year and be Year 2000 compliant, e.g., 30-06-1998; xx-08-1998; xx-xx-1998.
- **2. REPORT TYPE**. State the type of report, such as final, technical, interim, memorandum, master's thesis, progress, quarterly, research, special, group study, etc.
- **3. DATES COVERED**. Indicate the time during which the work was performed and the report was written, e.g., Jun 1997 Jun 1998; 1-10 Jun 1996; May Nov 1998; Nov 1998.
- **4. TITLE.** Enter title and subtitle with volume number and part number, if applicable. On classified documents, enter the title classification in parentheses.
- **5a. CONTRACT NUMBER**. Enter all contract numbers as they appear in the report, e.g. F33615-86-C-5169.
- **5b. GRANT NUMBER.** Enter all grant numbers as they appear in the report, e.g. 1F665702D1257.
- **5c. PROGRAM ELEMENT NUMBER.** Enter all program element numbers as they appear in the report, e.g. AFOSR-82-1234.
- **5d. PROJECT NUMBER.** Enter al project numbers as they appear in the report, e.g. 1F665702D1257: ILIR.
- **5e. TASK NUMBER.** Enter all task numbers as they appear in the report, e.g. 05; RF0330201; T4112.
- **5f. WORK UNIT NUMBER.** Enter all work unit numbers as they appear in the report, e.g. 001; AFAPL30480105.
- **6. AUTHOR(S).** Enter name(s) of person(s) responsible for writing the report, performing the research, or credited with the content of the report. The form of entry is the last name, first name, middle initial, and additional qualifiers separated by commas, e.g. Smith, Richard, Jr.
- 7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES). Self-explanatory.

- **8. PERFORMING ORGANIZATION REPORT NUMBER.** Enter all unique alphanumeric report numbers assigned by the performing organization, e.g. BRL-1234; AFWL-TR-85-4017-Vol-21-PT-2.
- 9. SPONSORING/MONITORS AGENCY NAME(S) AND ADDRESS(ES). Enter the name and address of the organization(s) financially responsible for and monitoring the work.
- **10. SPONSOR/MONITOR'S ACRONYM(S).** Enter, if available, e.g. BRL, ARDEC, NADC.
- **11. SPONSOR/MONITOR'S REPORT NUMBER(S).** Enter report number as assigned by the sponsoring/ monitoring agency, if available, e.g. BRL-TR-829; -215.
- 12. DISTRIBUTION/AVAILABILITY
 STATEMENT. Use agency-mandated
 availability statements to indicate the public
 availability or distribution limitations of the report.
 If additional limitations/restrictions or special
 markings are indicated, follow agency
 authorization procedures, e.g. RD/FRD,
 PROPIN, ITAR, etc. Include copyright
 information.
- **13. SUPPLEMENTARY NOTES.** Enter information not included elsewhere such as: prepared in cooperation with; translation of; report supersedes; old edition number, etc.
- **14. ABSTRACT.** A brief (approximately 200 words) factual summary of the most significant information.
- **15. SUBJECT TERMS.** Key words or phrases identifying major concepts in the report.
- **16. SECURITY CLASSIFICATION.** Enter security classification in accordance with security classification regulations, e.g. U, C, S, etc. If this form contains classified information, stamp classification level on the top and bottom of this page.
- 17. LIMITATION OF ABSTRACT. This block must be completed to assign a distribution limitation to the abstract. Enter UU (Unclassified Unlimited) or SAR (Same as Report). An entry in this block is necessary if the abstract is to be limited.

PRESTRESSED CARBON FIBER COMPOSITE OVERWRAPPED GUN TUBE

Andrew Littlefield and Edward Hyland US Army RDECOM-ARDEC Benét Laboratories Watervliet, NY 12180 USA

ABSTRACT

The emphasis on lightweight large caliber weapons systems has placed the focus on the use of advanced composite materials. Using composite materials not only directly removes weight from the gun tube but, by better balancing the tube, allows the use of smaller drive systems, thus further enhancing the system weight loss. Additionally the use of high stiffness composites helps with pointing accuracy and to alleviate the dynamic strain phenomenon encountered with high velocity projectiles.

Traditionally there were two issues with composite jackets: the coefficient of thermal expansion mismatch between the steel substrate and the composite jacket causing a gap, and the lack of favorable prestress in the jacket. Dealing with these issues greatly complicated the manufacturing process to the point where mass-producing the barrels would have been problematic at best. By using a thermoplastic resin, a "cure on the fly" process and winding under tension the manufacturability of the barrels has been greatly improved, the gap has been eliminated, and a favorable prestress has been achieved. This paper will present the design, manufacture and testing of a 120mm barrel utilizing this process with IM7 carbon fibers in a polyetheretherketone (PEEK) matrix.

TABLE OF CONTENTS

INTRODU	JCTION	<u>Page</u> 1			
	AND ANALYSIS				
MANUFACTURE					
NON-DESTRUCTIVE EVALUATION					
FIRING R	FIRING RESULTS				
CONCLU	CONCLUSION				
ACKNOW	/LEDGEMENTS	6			
REFERE	NCES	6			
	LIST OF FIGURES AND TABLES				
		<u>Page</u>			
Figure 1.	Dynamic FEA analysis of a steel tube with a composite jacket – Mises stress, 100x magnification	2			
Figure 2.	An axial ply being applied to the gun barrel	3			
Figure 3.	Modal testing setup	3			
Figure 4.	Acoustic Emission Test Setup	4			
Figure 5.	Test firing at APG	4			
Figure 6.	Type 1 - Experimental & Theoretical Strain vs. Time	5			
Figure 7.	Type 2 - Experimental & Theoretical Strain vs. Time	5			
Table 1.	Modal Impact Testing Results	4			

INTRODUCTION

Previous composite wrapped gun tube efforts have been under taken by Benét Laboratories during the late 1980's and early 1990's. These efforts led to the fabrication and test of several 105mm and 120mm gun tubes. An outcome of this work was the need to prevent or eliminate the formation of a gap, on the order of 0.1 mm (0.004 in), between the composite overwrap and gun steel liner during the composite curing process. The gap formed due to the coefficient of thermal expansion (CTE) mismatch between steel and composite. This gap effectively prevented or reduced the load carrying capability of the composite. To overcome the problem, the gun tube was autofrettaged (method of achieving compressive residual stresses at the bore by plastic deformation) after the application of the composite. The autofrettage effectively closed the gap, and also imparted some favorable residual stresses to the gun tube structure. There were, however, three problems with this approach; first, the thermal soak treatment used to stabilize the residual stresses in the tube after autofrettage could not be conducted. thermal soak is done at temperatures of 343 to 371 °C (650 to 700 °F) which is well above the maximum use temperature of the composite. The second problem was that the tube could not be chrome plated since the process requires the tube to be immersed in chromic acid, which would destroy the composite and contaminate the plating bath. The third problem is the creation of extremely high radial stresses at the steel / composite overwrap which may be higher than firing stresses (Parker et al., 2005).

One approach to solving these problems was the 105mm Multi-Role Armament and Ammunition System (MRAAS) Swing Chamber Launcher (Littlefield and Hyland, 2002). In this case the CTE mismatch was handled by tailoring the lay-up. A combination of fiberglass and graphite was used with the ply angles being adjusted such that the lay-up's CTE matched that of the steel. This resulted in no gap forming between the composite and the steel but the performance of the composite was not optimum.

The composites used on these efforts were all thermoset materials; therefore the curing process took place after composite wrapping. For the current Advanced Technology Demonstration (ATD) effort, thermoplastic composites will be used. The advantage of thermoplastics is that they do not need a cure cycle but rather can be melted and recrystallized/consolidated immediately after being placed on the gun tube. This results in a "cure in place" type fabrication technique. Heating of the

composite is localized, minimizing heat input to the composite and gun tube. This process mitigates thermal expansion effects and effectively eliminates the gap problem. The composite can therefore be placed onto the gun tube after the autofrettage thermal soak and chrome plate application.

One of the challenges of the composite wrapped gun tube will be to handle the dynamic loading environment of a gun tube. Firing data of gun tube strain have shown that the measured strains are typically higher than expected from static ballistic This increase in tube strain is pressure alone. attributed to both the loading condition, which is effectively a square wave, as well as high speed dynamic loading of the gun tube during projectile passage. In most cases, this strain is typically 8-10% above the statically predicted (open ended cylinder, Lame equations) values. In situations where thin walled gun tubes and high velocity projectiles are used, the strains can be significantly higher, on the order of 300-400%. This phenomenon is known as gun tube dynamic strain and has been an area of study for many years by Benét Laboratories (Simkins, 1987; Hasenbein et al., 1990; Hasenbein et al., 1992). In the development of the Light Weight 120mm (LW120) cannon, this phenomenon will be of special interest since the LW120 will be have a thinner tube wall than the current 120mm M256 cannon and thus it will be more prevalent.

The 120mm Line of Sight / Beyond Line of Sight (LOS/BLOS) ATD is tasked to design, develop & demonstrate new armament & ammunition technologies for use in the Army's Future Combat System (FCS). The specific role the ATD plays is to support the development of the main armament for the Mounted Combat System (MCS), which will be equipped with a 120mm main armament and will provide Line of Sight and Beyond Line of Sight firing capabilities.

One of the tasks assigned to the 120mm LOS/BLOS Gun Assembly Team was to provide a light weight 120mm gun assembly for the MCS vehicle. The focus of this report is the use of an organic composite overwrap to lighten the weight and reduce the imbalance of the gun tube. The ATD is scheduled to deliver two prototype composite wrapped gun tubes. The first tube, Serial No. ATD-1, was the first large caliber gun tube to be wrapped with thermoplastics and was reported on previously (Littlefield et al., 2006). This report will focus on the 2nd of these tubes, Serial No. ATD-3. In this second tube the thermoplastic is applied under tension to induce a favorable prestress in the composite jacket.

DESIGN AND ANALYSIS

Initially a lightweight all steel 120mm gun tube was designed using traditional methods. The steel design had a weight of 889 kg and was 5460 mm in length. The goal of the composite design was to match or exceed the frequency of the first bending mode of the steel design as well as match the residual hoop stress distribution through the gun tube wall, while saving weight.

Thermoplastic composites were used instead of thermosets in order to take advantage of the "cure in place" fabrication technique. Additionally applying the composite under tension helped to build in a favorable prestress in the composite jacket. Besides this manufacturing consideration the composite overwind had to be able to withstand the significant forces and heat fluctuations associated with firing the weapon.

IM7 fiber with a polyetheretherketone (PEEK) matrix was the material selected for this project for several reasons. The first is the superior strength (2.07 GPa (300 ksi) in the fiber direction), modulus (138 GPa (20 msi) in the fiber direction) and toughness of the composite when compared to the majority of thermoset and other thermoplastic materials. The second reason for the selection of this material was its high melt point (653 °F / 345 °C). The final reason for the selection of this material was its excellent chemical resistance; in particular, its resistance to petrochemical fluids that would be encountered in the day to day operation of a large The cost of thermoplastics, while in machine. general higher than thermoset counterparts (~20%), was offset by the fact that there would be no autoclave post cure required. With a shape as complex and large as this, bagging and autoclaving add significant expense (up to 20%) to thermoset processing, plus the capital investment in a large autoclave (approx \$300,000 for one large enough to process this gun tube), making thermoplastics a competitive alternative.

The tube's natural frequency (especially the first bending mode) affects the gun aiming and stabilization system. Maintaining the same natural frequency as an all steel version of the gun tube minimizes changes to these systems. In addition, if the natural frequency gets too low, it may approach the natural frequency of the riding loads of the vehicle. Excitation of the natural frequency may then occur leading to a condition in which stabilization of the gun tube becomes impossible.

Large caliber gun tubes often use autofrettage to impart favorable residual stresses into the gun tube structure. Since we were replacing some of the steel with composites, it was vital that the composite provide the same residual stress distribution as the original steel. To accomplish this, the residual stress distribution through the tube wall, including autofrettage and the composite wrap were modeled.

Static, normal mode and dynamic analyses were all performed. For the dynamic analysis, a pressure load was moved down the bore of the tube to simulate a projectile. A graphical result of this analysis can be seen in Figure 1.

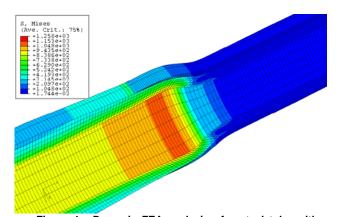


Figure 1. Dynamic FEA analysis of a steel tube with a composite jacket – Mises stress, 100x magnification.

These analyses were repeated until a lay-up was arrived at that met or exceeded all of the metrics. The final lay up consisted of a mixture of hoop and axial plies. The hoop plies were to be wound under tension to match the residual stress distribution of the original all steel design. Two ±45 degree layers of S2/PEEK were added on the outside to protect the carbon fiber layers. This lay-up resulted in 113.4 kg (250 lbs) of steel being removed and 20.4 kg (45 lbs) of composite being added for a net weight savings of 93 kg (205 lbs).

MANUFACTURE

The steel portion of the gun barrel was manufactured according to the normal process, except that an area was undercut for the composite.

The composite was applied utilizing a robotic fiber placement process to precisely place and consolidate strips of thermoplastic prepreg tape. The process uses a hot gas torch (HGT) to melt the prepreg and then consolidates it with a pressure roller. Throughout the process the tape is held under

tension and upon cooling this tension is locked in; inducing a residual stress into the part.

There were three major issues that needed to be overcome in order to fabricate the overwind:

- Tightness of fit between overwrap and barrel
- Galvanic corrosion between overwrap and barrel
- Maintaining the desired outside diameter (OD)

Winding under tension helps to ensure a tight fit between the overwrap and barrel but beyond this it was decided to cool the barrel, thus causing it to shrink during processing. Upon returning to room temperature the barrel attempts to grow in size but is constrained by the composite. In this way we are using the CTE mismatch to help form a tighter fit between the steel and composite instead of a gap, as was the case in older thermoset overwrapped gun tubes. This cooling process was found to induce level of residual stress equivalent to approximately 133 N (30 lbs) of winding tension.

Additionally the cooling helps to remove the heat generated from the fiber placement process. Without cooling the barrel temperature would have quickly heated to between 60 and 65 °C (140 to 150 °F). The exact temperature the barrel was cooled to can not be released but it was within the operational temperature of the gun system so it will not adversely affect the mechanical properties of the steel.

If carbon fiber is brought into direct contact with steel galvanic corrosion would take place. To avoid this two layers of S2 fiberglass / PEEK were placed between the steel and the carbon fiber. This thin layer is enough to act as an insulator but think enough to not effect the performance of the overwrap.

Due to some standard variation in raw material thickness (specification for the material allows a +/-0.0127 mm variation in tape thickness), close attention was paid to the OD during fabrication. Modifications to ply lengths and locations were made to maintain the desired final OD.

Figure 2 shows an axial ply being applied to the gun barrel. The white area is frost that develops on the part due to the chilling of the barrel. The hot gas torch vaporizes this as it applies the tape, so that none of the moisture finds it way into the part.



Figure 2. An axial ply being applied to the gun barrel.

NON-DESTRUCTIVE EVALUATION

Modal impact, pressure, and acoustic emission (AE) testing were all performed to assess the state of the composite overwrap. This was done both before and after test firing to assess if the firing had any detrimental effects. Ultrasonic inspection was planned if any of the tests uncovered possible areas of damage.

Modal impact testing was performed both prior to and after applying the composite to determine the effect of the overwrap on tube stiffness. In all cases the tube was hung from springs to simulate free-free boundary conditions. This setup can be seen in Figure 3.



Figure 3. Modal testing setup.

Accelerometers were placed at the muzzle and every foot (304.8 mm) down the length of the composite. The tube was then impacted 219 mm from the muzzle and the response of the accelerometers was recorded. After this, all but the muzzle accelerometer were removed and the tube was then impacted at each previous accelerometer location.

The results of this testing for the first three modes can be seen in Table 1. The composite wrap

slightly increased the stiffness of the gun. These results were compared to the FEA analysis and were found to be in good agreement. Not only did this result help to validate the FEA models but also ensured that energy was being transferred from the composite to the steel and vice versa. The slight drop in frequency after firing was determined to be within acceptable test error.

TABLE 1. MODAL IMPACT TESTING RESULTS

	Mode (Hz)		
	First	Second	Third
Before Wrap	26.50	81.00	174.00
After Wrap	28.75	85.25	178.75
After Firing	28.25	83.50	173.75

The pressure and AE tests were conducted at the same time as they both required pressurizing the gun tube. The pressure test helps to ensure that there is no gap between the steel and the overwrap. If a gap exists then there would be a delay in the composite picking up the pressure load applied to the bore. For the AE test the tube is pressured twice. The first time there will be some fiber and matrix cracking as any defects need to work themselves out. The second loading should be quiet. If the second loading produces any noise events they could be an indication of damage and need to be investigated.

Standard rosette strain gages were placed at two axial locations along the length of the composite. At each location a gage was placed at the 12, 3, 6 and 9 O'clock positions. The gauges were oriented to record both hoop and axial strain. These same gauges were later used in the firing test. A mandrel was then inserted into the bore under the composite and was pressurized to 68.9 MPa (10 ksi). The strain readings were recorded every 6.89 MPa (1000 psi).



Figure 4. Acoustic Emission Test Setup.

Eight Physical Acoustics R-151 acoustic emission sensors were set up in an F-array so that the location of any suspected damage could be located. The mandrel used to pressurize the tube was only 1828.8 mm (72") in length so the pressure/AE test was conducted twice to cover the entire length of the composite. Figure 4 shows the setup of the AE sensors for the second test area. The pressure data that was collected were within 3% of the FEA predictions. The post firing test showed no signs of degradation due to firing.

FIRING RESULTS

In December 2004, May 2005 and July 2005 the gun was taken to Aberdeen Proving Ground (APG), MD for test firing. The gun was fired in direct and indirect fire modes though strain data was collected for only the first 20 direct fire shots. During these shots a series of two round types were fired. Figure 5 shows a direct fire shot.

The test instrumentation used was standard rosette strain gauges. Gauges were placed at two axial locations along the composite area of the tube. At each axial location a gage was placed at the 12, 3, 6 and 9 o'clock positions. Measurements of axial and circumferential (hoop) strain were recorded throughout the first 20 rounds of the test.



Figure 5. Test firing at APG

Table 2 gives both the theoretical and experimental strains for the two round types fired. Looking at the table it can be seen that there is good qualitative and quantitative agreement between theoretical and measured strain levels. The response for the round type 1 was higher than expected but this is believed to be due to higher than expected pressures generated by the round. The results for round type 2 (the worst case round) were

excellent with test results at both locations within 3% of theoretical.

TABLE 2. EXPERIMENTAL AND THEORETICAL HOOP STRAINS (ME)

STRAINS (ME)							
Round Type	#1	#2					
Location 1 Experimental	Mean 1755 Std Dev 33	Mean 1766 Std Dev 86					
Location 1 Theoretical	1527	1719					
Location 2 Experimental	Mean 2160 Std Dev 145	Mean 1933 Std Dev 289					
Location 2 Theoretical	1575	1922					

Figure 6 and Figure 7 show the experimental and theoretical strains vs. time at axial location 1 for both round types. Looking at the figures it can be seen again that there is good agreement between theoretical and experimental results.

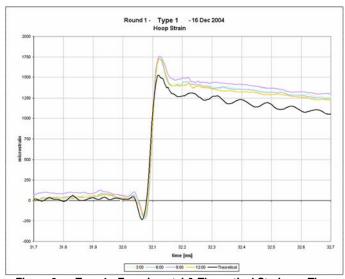


Figure 6. Type 1 - Experimental & Theoretical Strain vs. Time

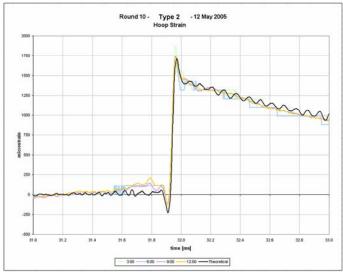


Figure 7. Type 2 - Experimental & Theoretical Strain vs. Time

CONCLUSION

A lightweight composite wrapped 120mm gun tube was successfully designed, manufactured, and test fired. A thermoplastic matrix was used, allowing for cure in place fabrication. This avoided the manufacturing complications due to coefficient of thermal expansion mismatch encountered in previous attempts at composite wrapped gun tubes. The prepreg was applied under tension resulting in a favorable prestress in the composite jacket. The design resulted in a gun tube that was 93 kg (205 lbs) lighter than its all steel counterpart while maintaining the same first bending mode and cross sectional profile.

Finite element models were used to help predict the response of the gun tube to firing loads. These models were validated through non-destructive testing and later shown to be in good agreement with the firing results. The composite jacket survived the firing with no apparent damage.

Overall, this effort was very successful and the data collected will be very useful in the design of future composite wrapped gun tubes.

ACKNOWLEDGMENTS

The authors would like to thank the entire LOS/BLOS ATD team; the firing range crew at APG; the group at ADC that helped manufacture the tube; and Physical Acoustics for helping to perform the acoustic emission testing. Without the help of all these people this effort could never have been completed.

REFERENCES

- Hasenbein, R., Gabriele, A., Artus, B., Cunningham, G., and Gast, R., 1990, "Dynamic Strain Waves a Development Perspective," ARDEC Tech. Rep. ARCCB-TR-90030, Benét Laboratories, Watervliet, NY, 24 pp.
- Hasenbein, R., and Hyland, E., 1992, "Dynamic Strain Waves and Permanent Bore Enlargement," ARDEC Tech. Rep. ARCCB-TR-92042, Benét Laboratories, Watervliet, NY, 41pp.
- Littlefield, A., and Hyland, E., 2002, "Use of Composites on the FCS-MRAAS Swing Chamber Launcher for Reduced System Weight," 23rd Army Science Conference, Orlando, FL, CD-ROM, CP-08.
- Littlefield, A. Hyland, E., Andalora, A., Klein, N., Langone, R., and Becker, R., "Carbon Fiber / Thermoplastic Overwrapped Gun Tube," *ASME Journal of Pressure Vessel Technology*, **128**, 257-262, 2006.
- Parker, A.P., Troiano, E., and Underwood, J.H., "Stresses Within Compound Tubes Comprising a Steel Liner and an External Carbon-Fiber Wrapped Laminate", *ASME Journal of Pressure Vessel Technology*, **127**, 26 - 30, 2005.
- Simkins, T.E., 1987, "Resonance of Flexural Waves in Gun Tubes," ARDEC Tech. Rep ARCCB-TR 87008, Benét Laboratories, Watervliet, NY, 57 pp.